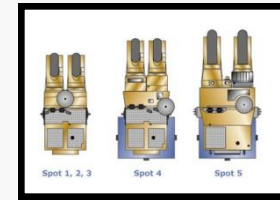
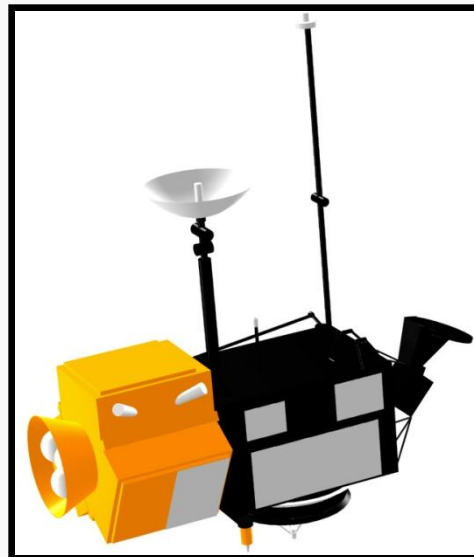
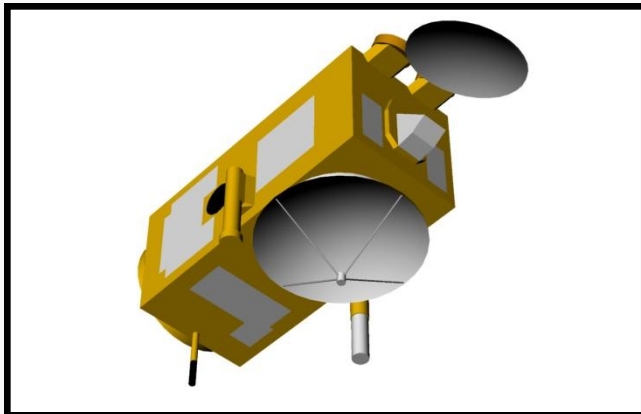




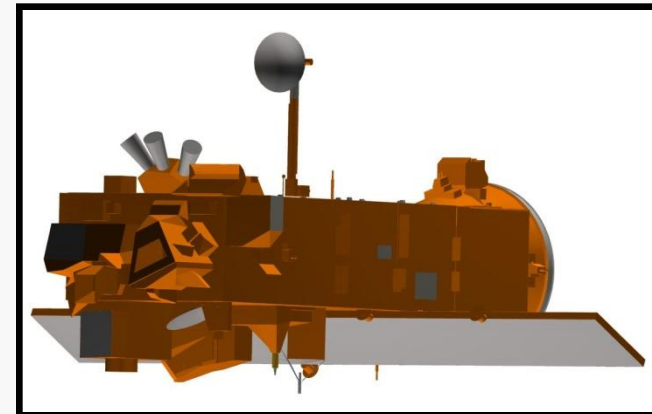
Non-Conservative Force Modelling for DORIS Satellites at UCL



Ant Sibthorpe

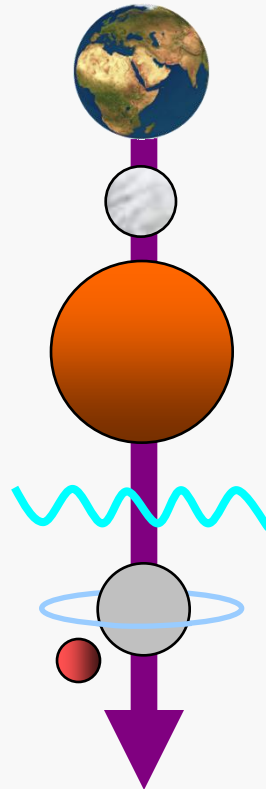


Marek Ziebart



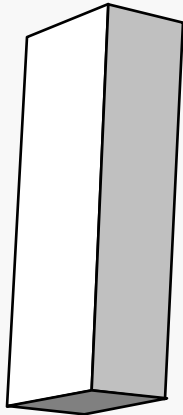
Forces acting on a spacecraft

- Earth gravity
- Lunar gravity
- Solar gravity
- Solar radiation pressure
- Atmospheric Drag
- Thermal forcing

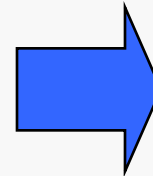
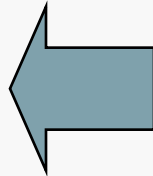


- Planetary Radiation Pressure
- Tidal effects
- Antenna thrust
- Planetary gravity & general relativistic effects

Modelling Satellite Surface Forces



Conventional approach



UCL approach

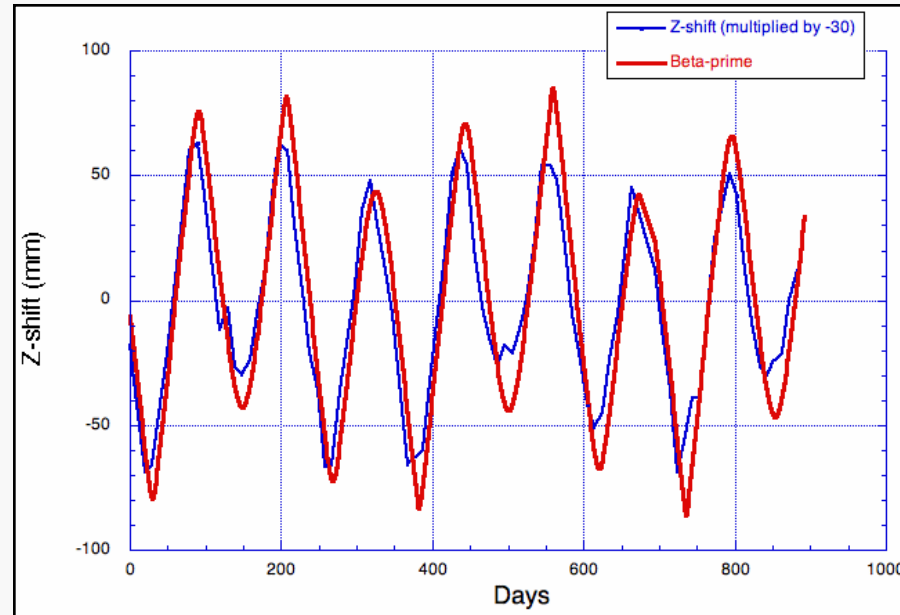
‘Box and wing’ model, rely upon empirical parameters to ‘Soak up’ mismodelling.

OR

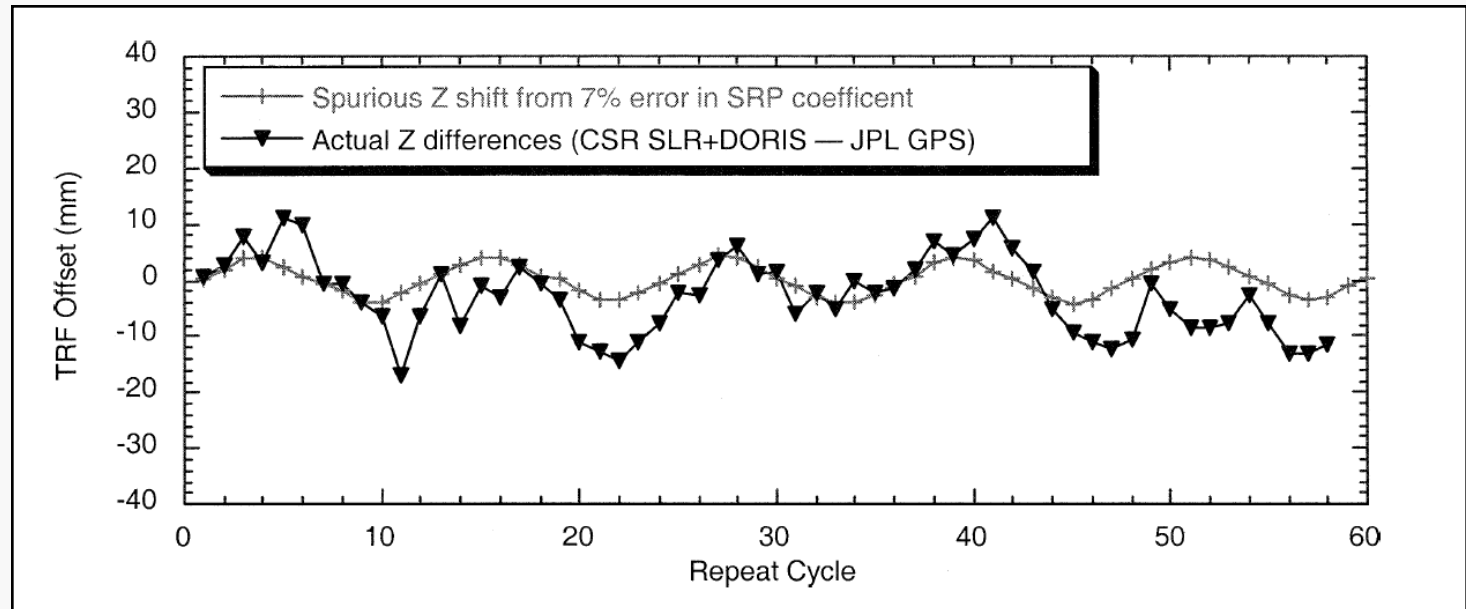
- Base modeling on physics and engineering.
- Just get it right.

Scale Factors

- Cr correlated with other parameters
- Scale errors cause spurious Z-axis variations in orbit

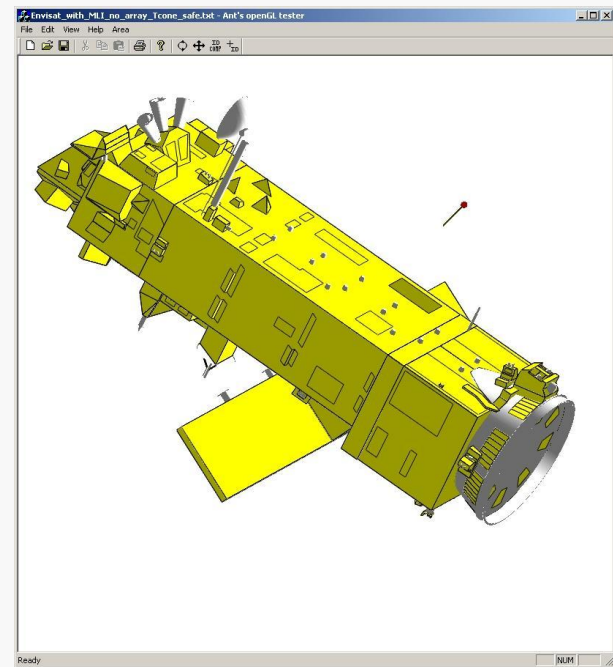
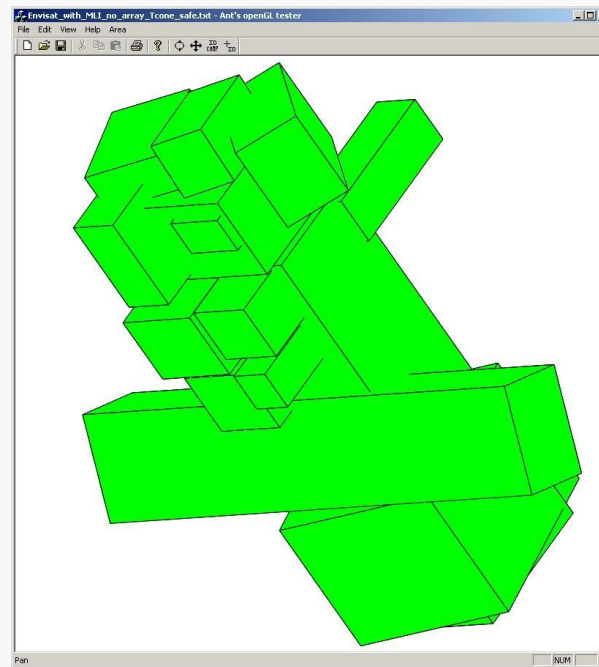
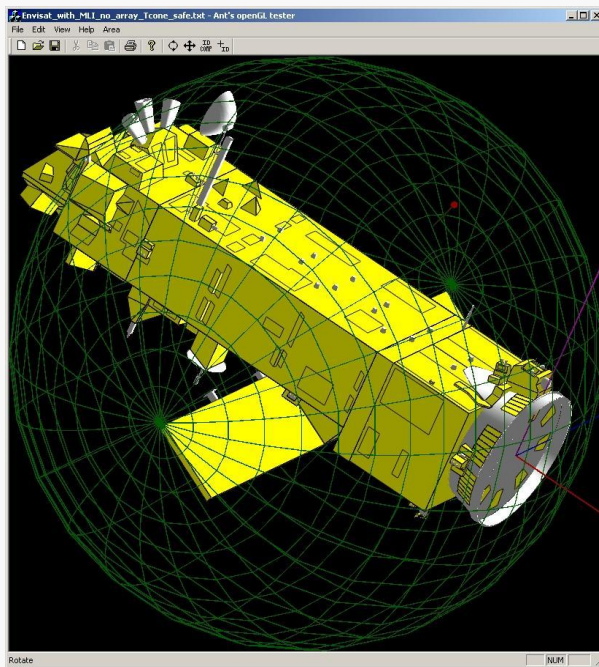


JASON-1 tests using different (3%) solar radiation pressure scale factors (John Ries, CSR)

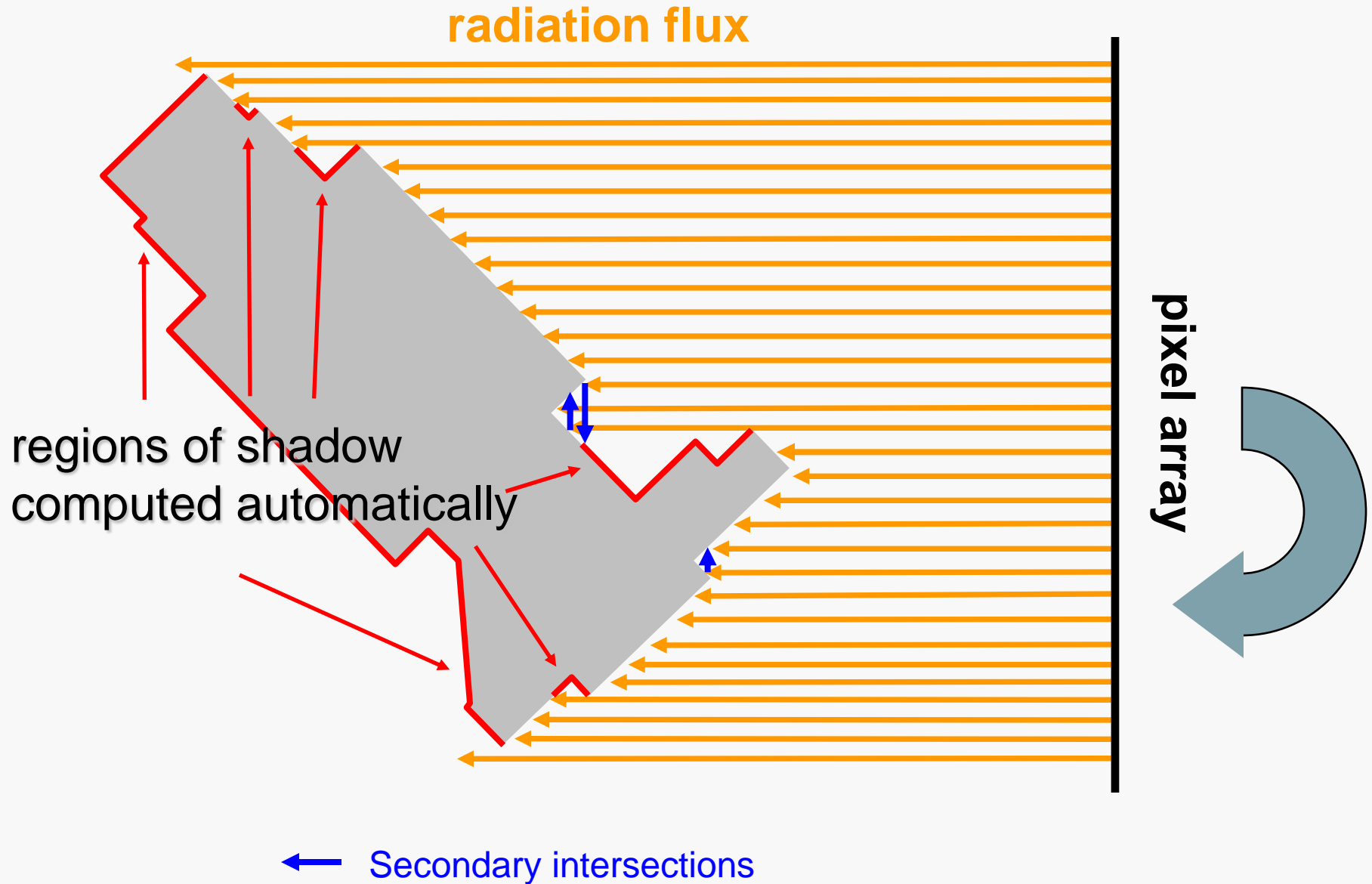


JASON-1 tests using different (7%) solar radiation pressure scale factors (Bruce Haines, JPL)

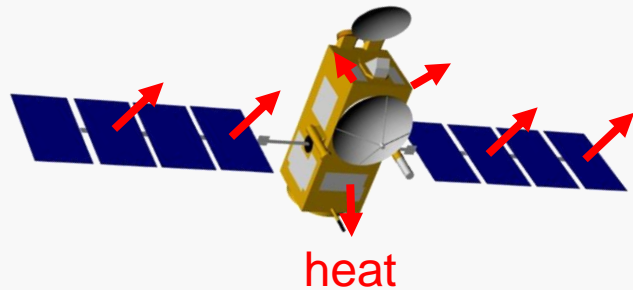
UCL Quality Control: View Tools



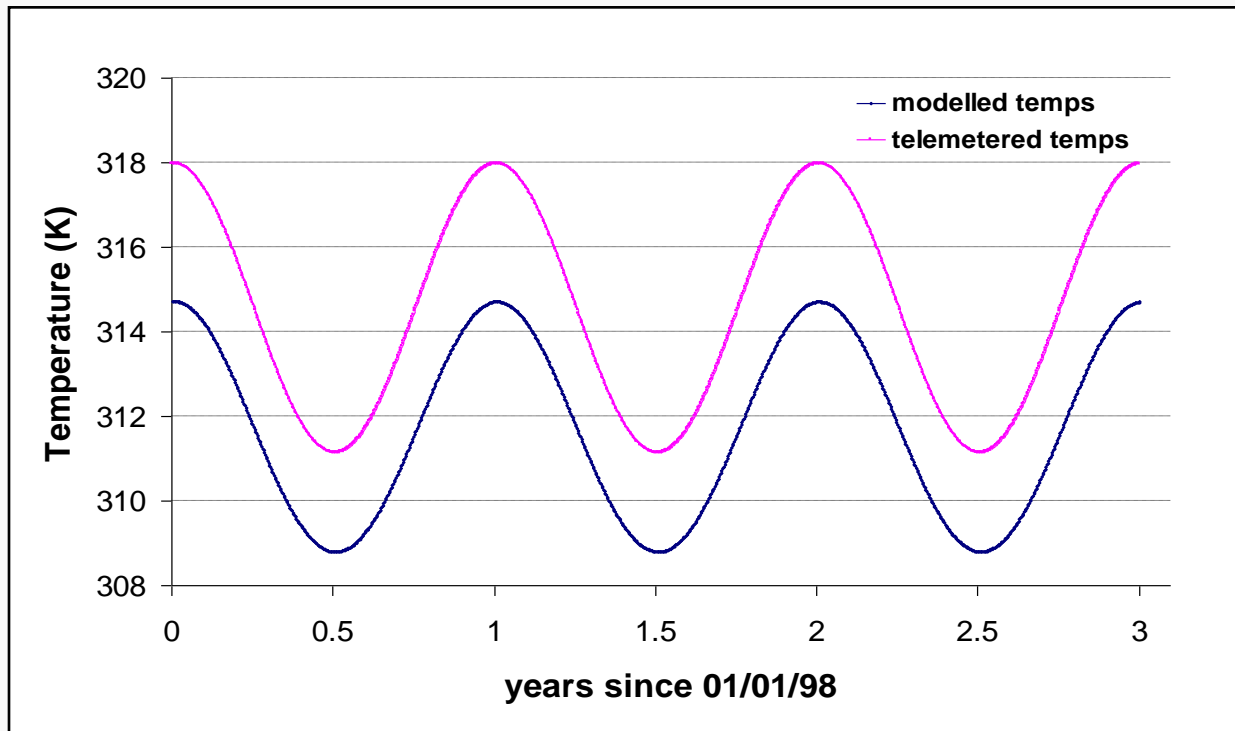
Ray Tracing



Thermal Modelling



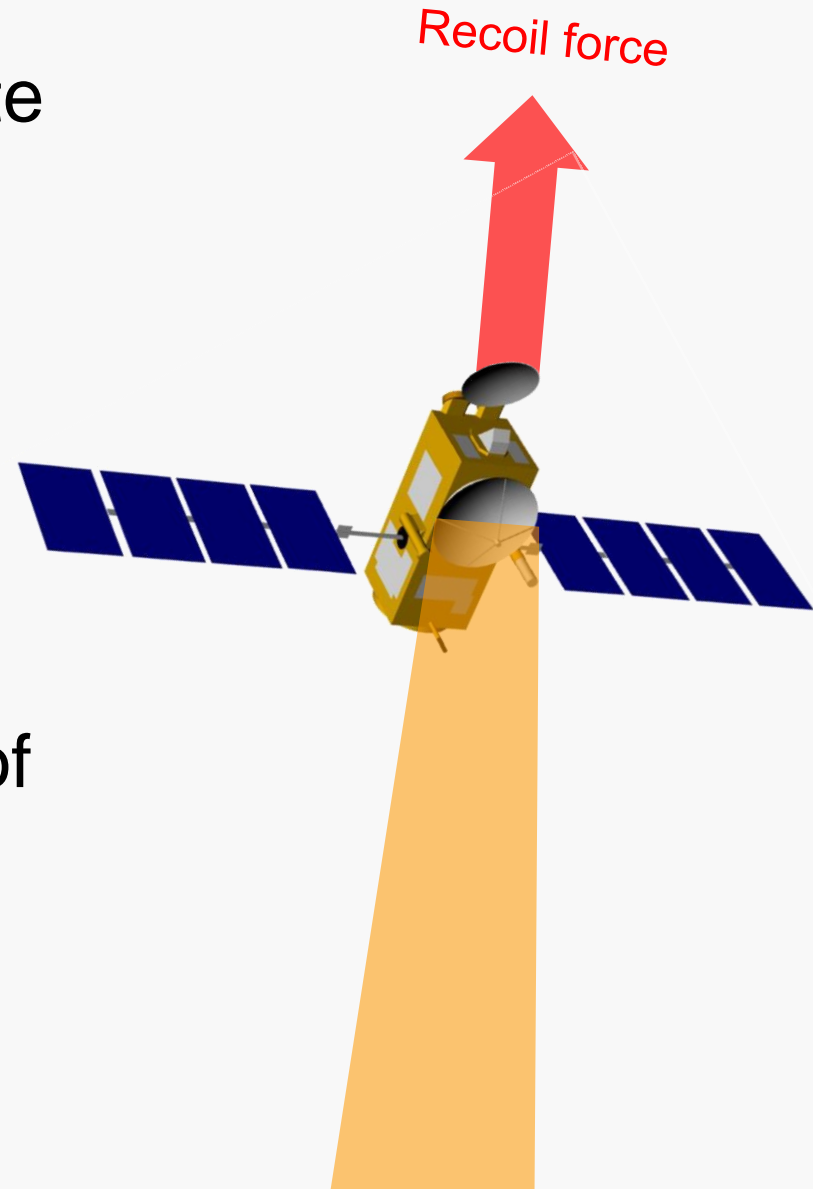
Anisotropic thermal emission from spacecraft results in a net acceleration



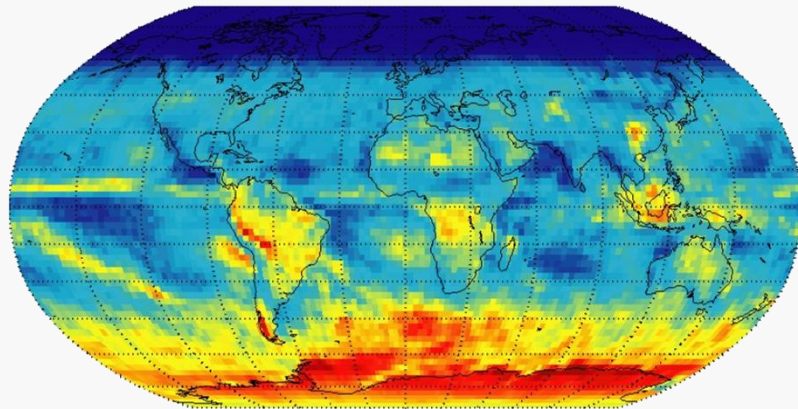
**Temps:
modelled vs
telemetered**

Antenna Thrust

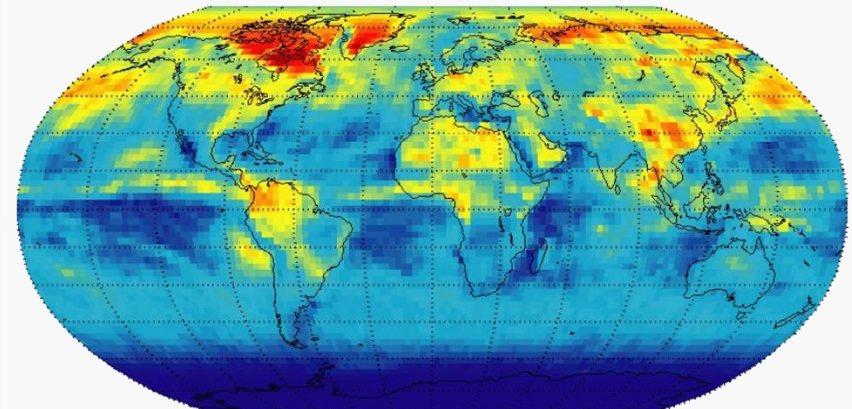
- Recoil force on satellite due to transmitted navigation, SAR & altimetry signals etc.
- Systematic and observable effect
- Requires knowledge of power transmission of satellites



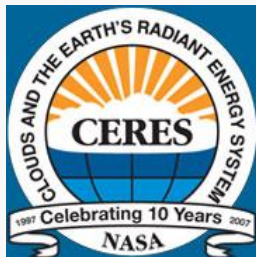
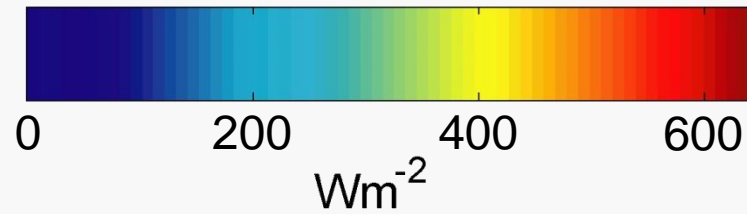
PRP: radiation complexity



Maximum SW: Dec 2003

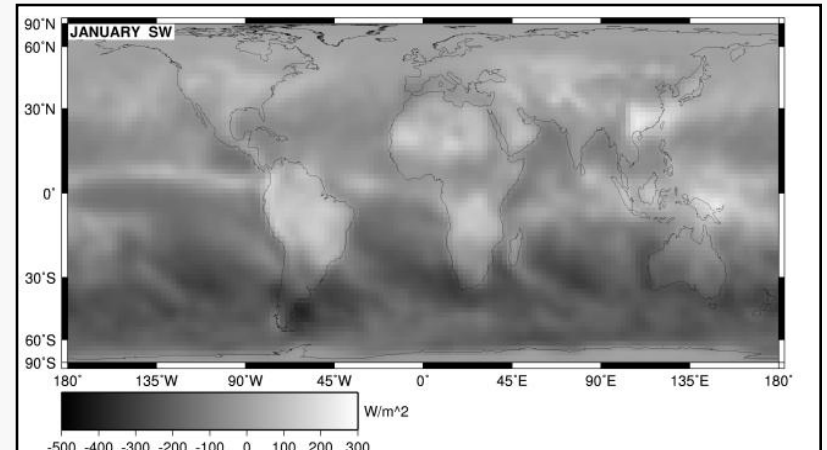
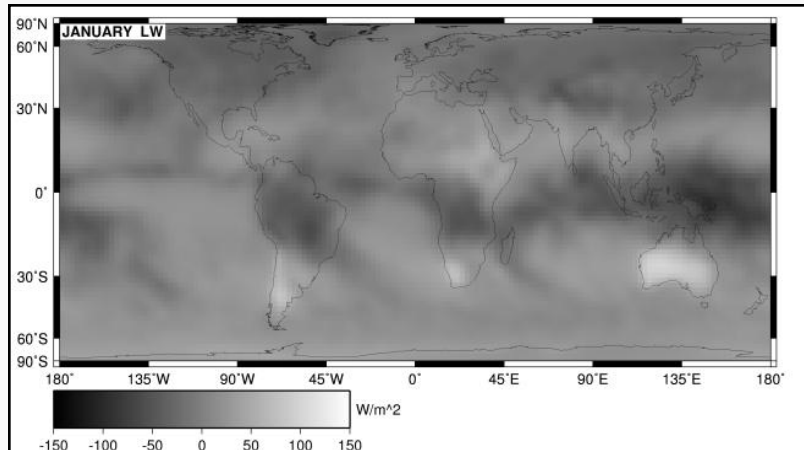
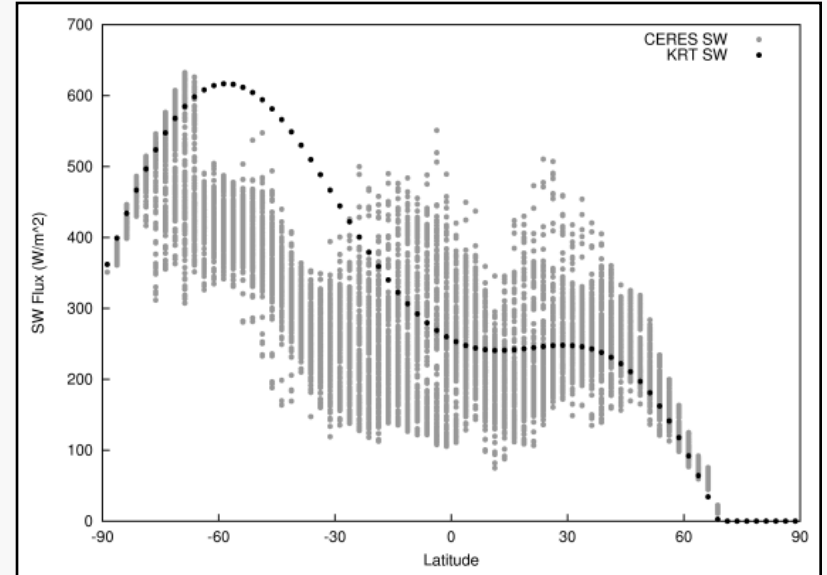
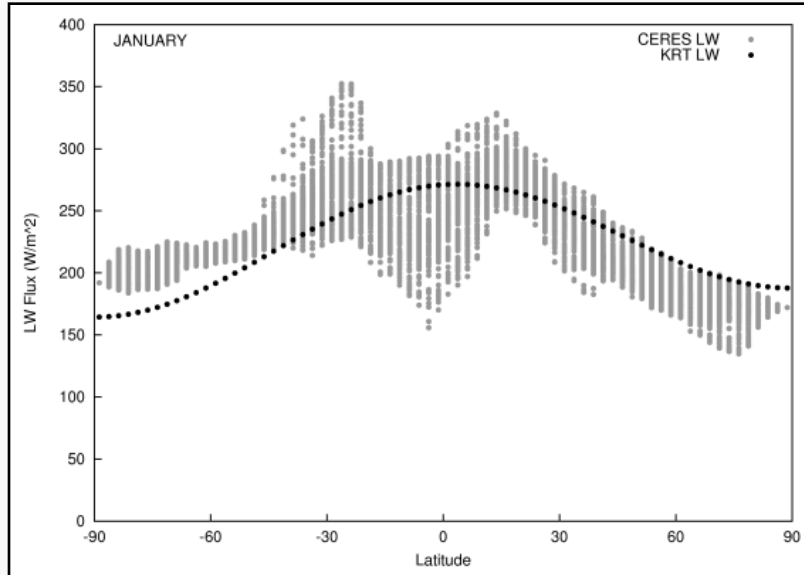


Maximum SW: May 2004



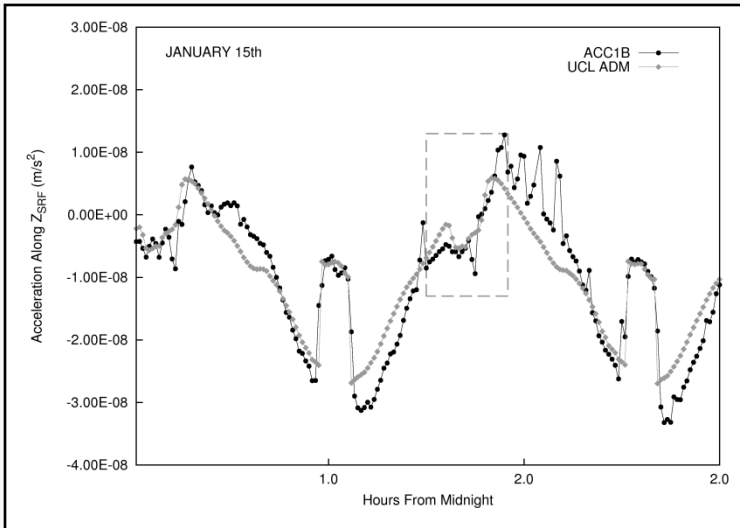
Radiant exitance: top of atmosphere

Radiant Exitance: CERES Vs KRT. January 15th, 2005.

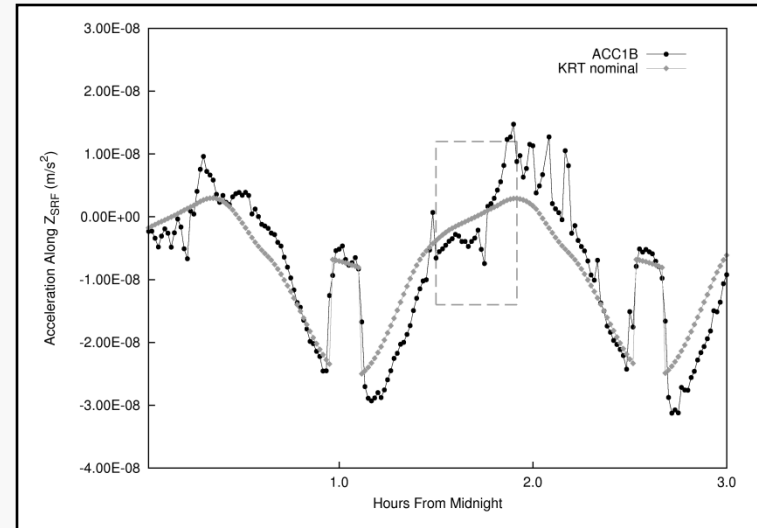


Radiant Exitance: CERES – KRT. January 15th, 2005.

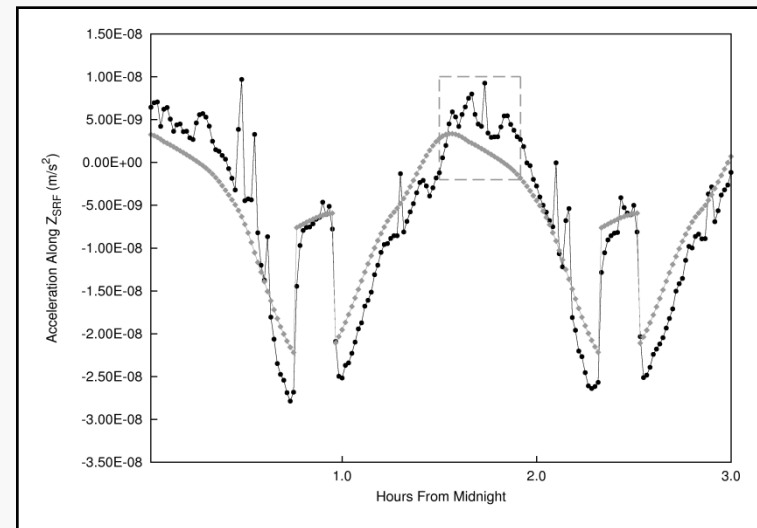
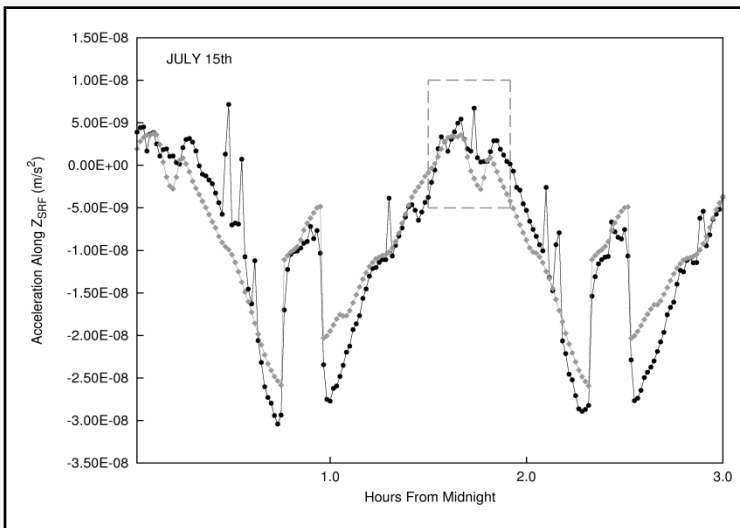
SRP + (UCL & Knocke) Vs GRACE ACC1B_Z



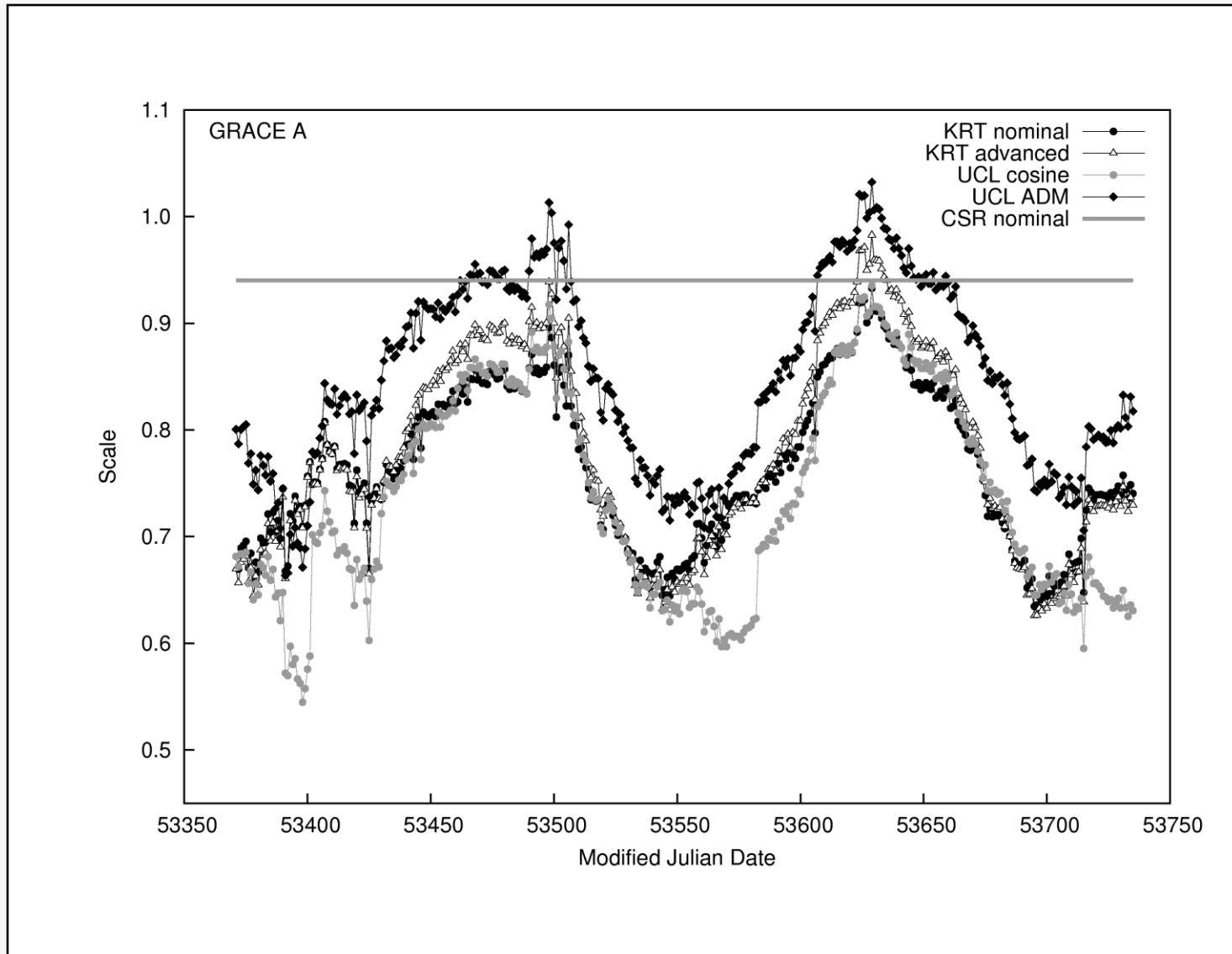
UCL: ADM



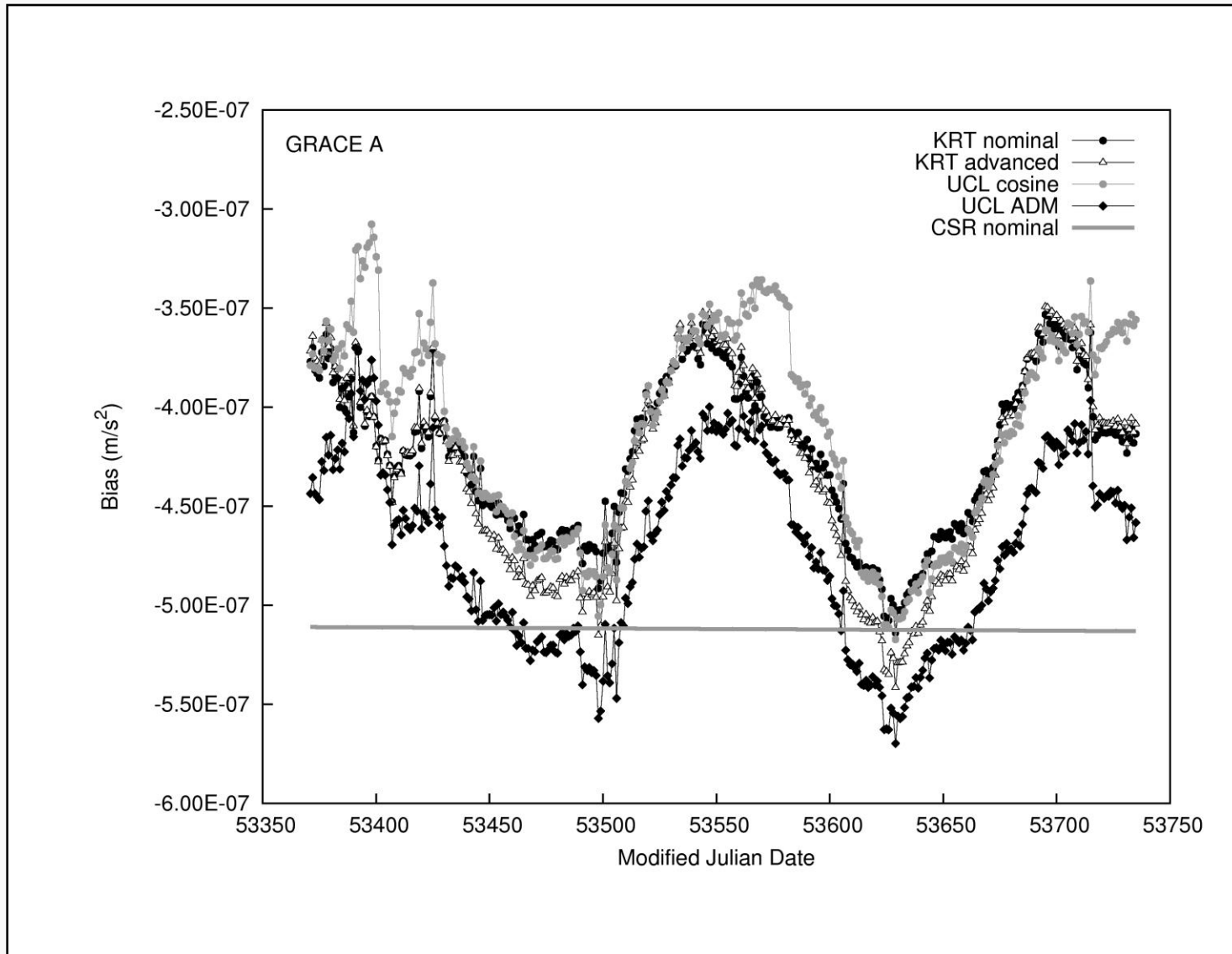
Knocke: Diffuse



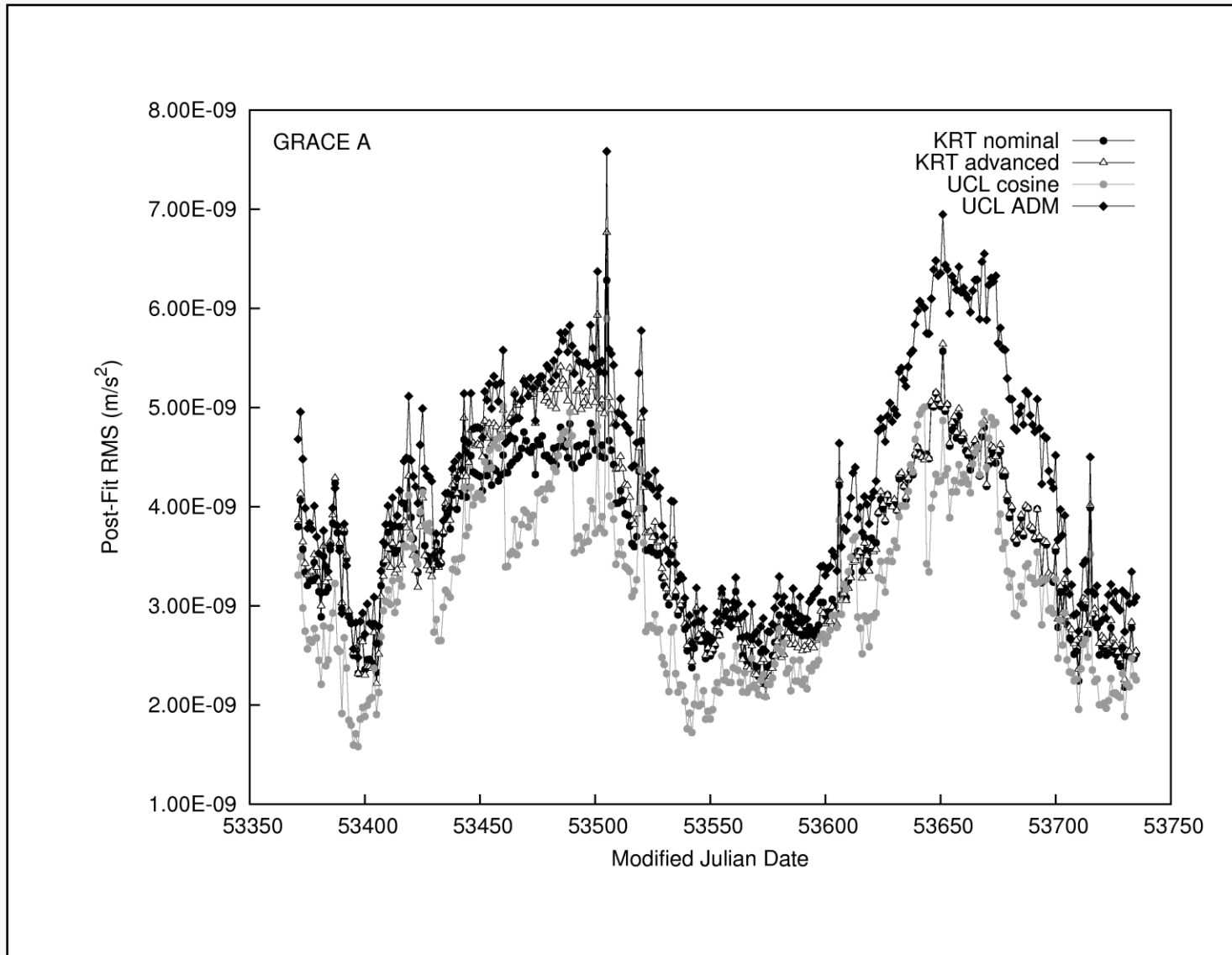
Scale to fit radial SRP+PRP to GRACE ACC1B_Z in 2005



Bias to fit radial SRP+PRP to GRACE ACC1B_Z in 2005

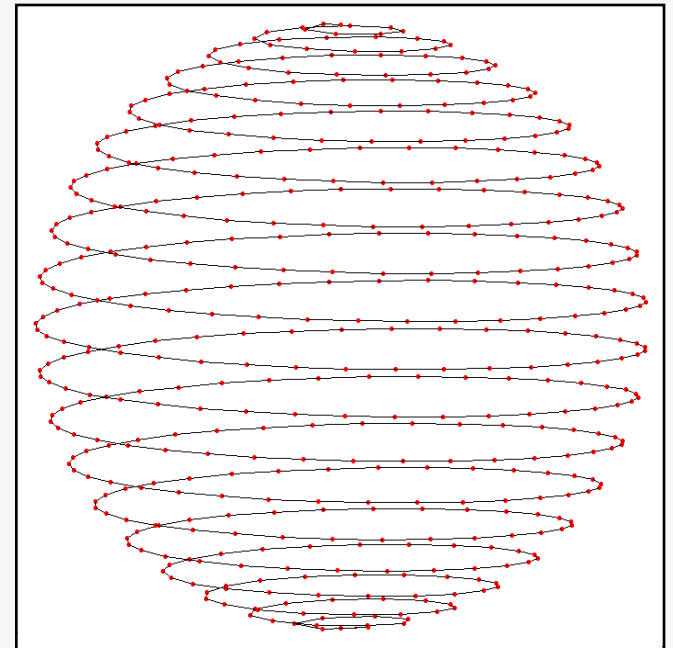


Post-fit RMS of radial SRP+PRP and GRACE ACC1B_Z, 2005

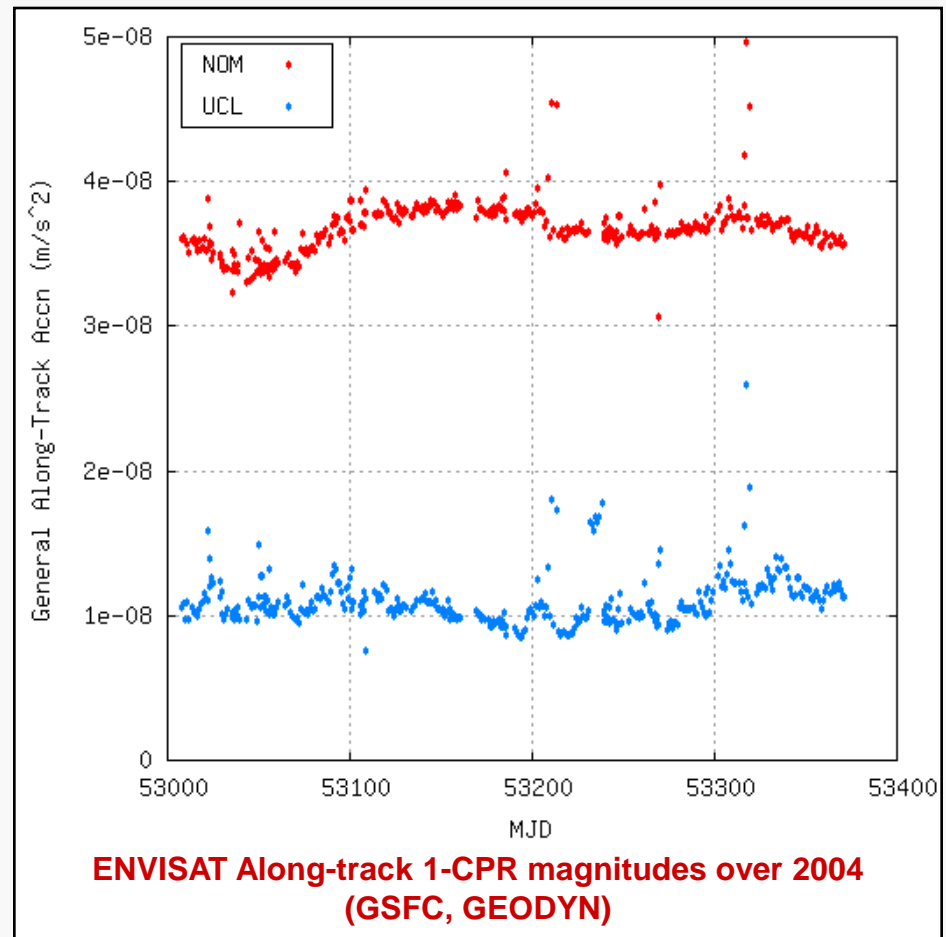
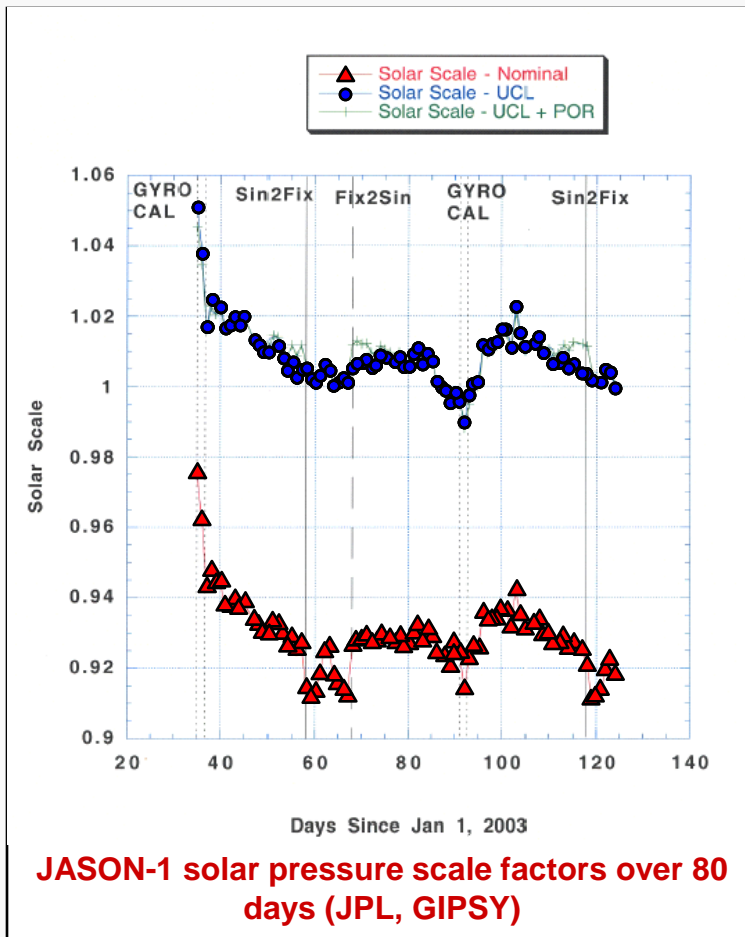


Simplicity Captures Complexity

- Geometry sampled discretely by placing pixel array at points in a plane or along a spiral.
- Fourier series or surface fitted to resulting accelerations using custom interpolation
- Final output is a Fourier series or a grid file to be interpolated bi-linearly in an integrator



UCL Model Results: JASON-1 & ENVISAT



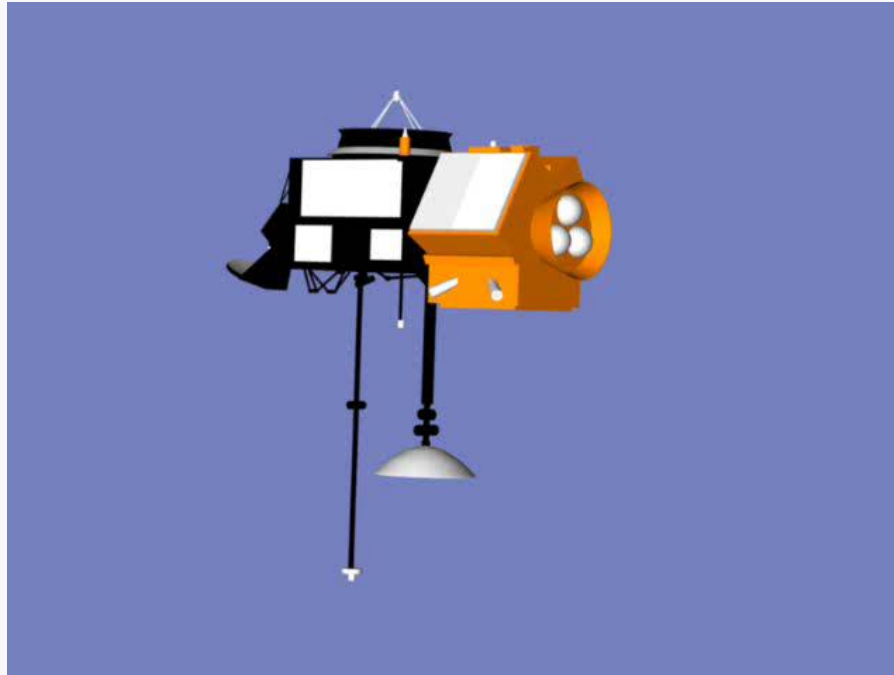
- UCL model $Cr \sim 1.01$
- More stable behaviour
- Macromodel $Cr \sim 0.93$

- Nominal Box-Wing and UCL ENVISAT model performance over 2004, $Cr = 1$

Conclusions

- Simplified satellite surface force models lead to reference frame and orbital biases
- UCL models significantly mitigate these errors without using empirical terms
- UCL modelling embraces complexity in spacecraft structure and environmental data and are accurate, fast and efficient when implemented
- This is still an active research area, and we hope to continue working on a range of satellite missions such as SPOT & TOPEX
- **Better modelling technologies now exist. They can, and should, be used**

UCL TOPEX Geometry



Thankyou

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