

# DORIS

DAYS  
2025

November, 3 and 5

**DORIS**  
**observations**

November 5, 2025

Session 2 – Presentation 1

DORIS Days 2025

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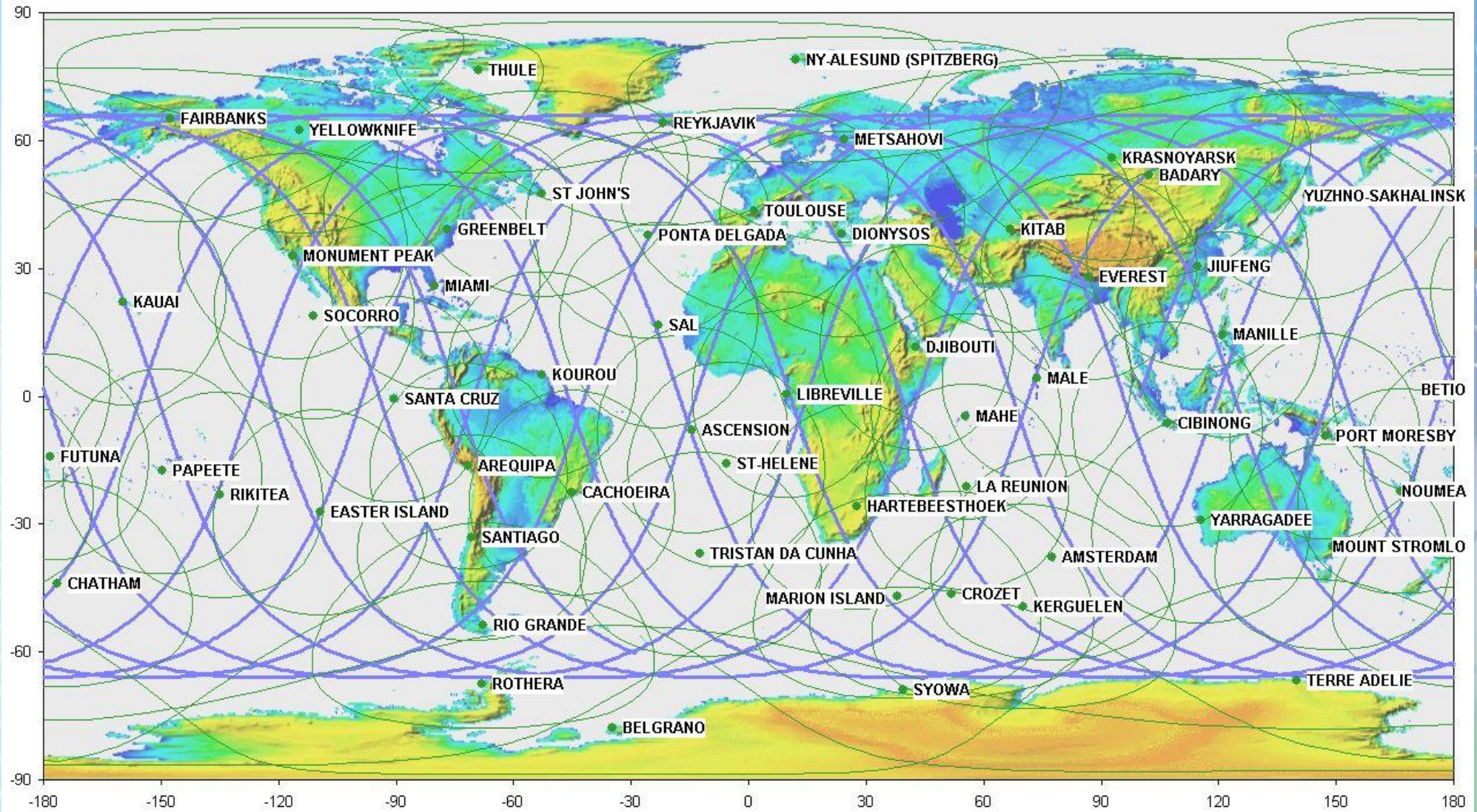
# Setup of this class

First: What is DORIS observing? Define the main topics that you need to consider (~25 min)

Second: Use the Jupyter notebook, run a simulation and browse through a DORIS Rinex file (~60 min)



## Jason-1 DORIS stations visibilities Elevation 12°

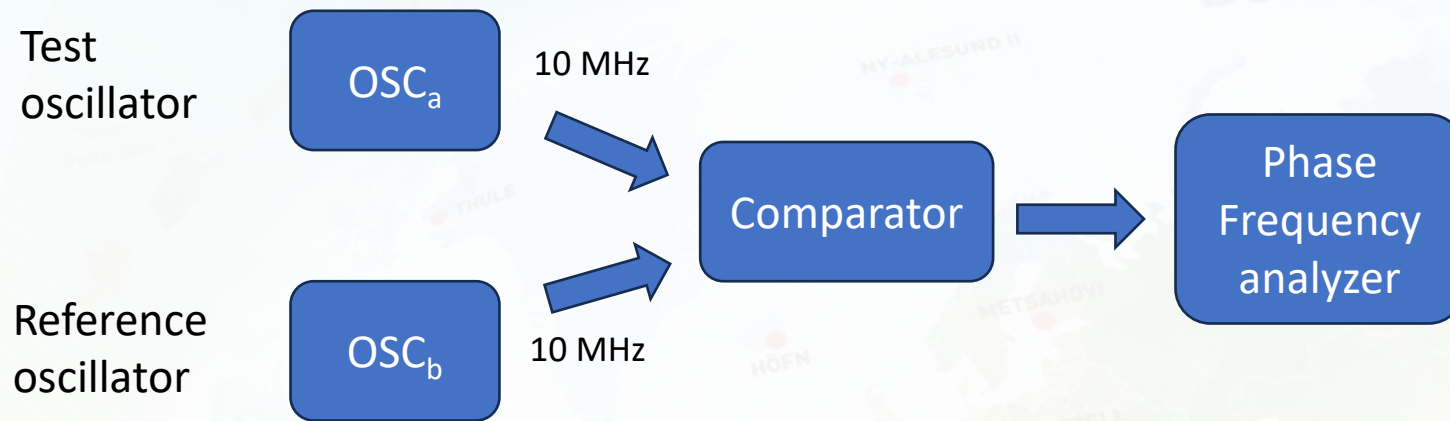




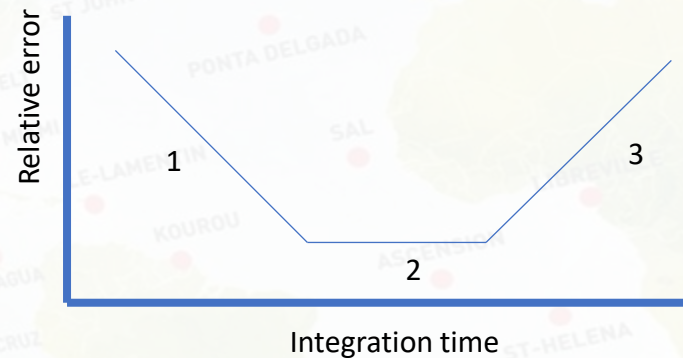
# Content

- Twin oscillators
  - Experiment in the laboratory
  - Experiment in the field
- Conclusions from the field setup
  - Range and velocity
  - Time synchronization
  - Light time correction and relativity
  - Refraction
  - Antenna's
- Hands on
  - Differences between DORIS and GNSS
  - Argos simulator experiment (via Jupyter notebook)

# Lab setup twin oscillators



Allan deviation analysis



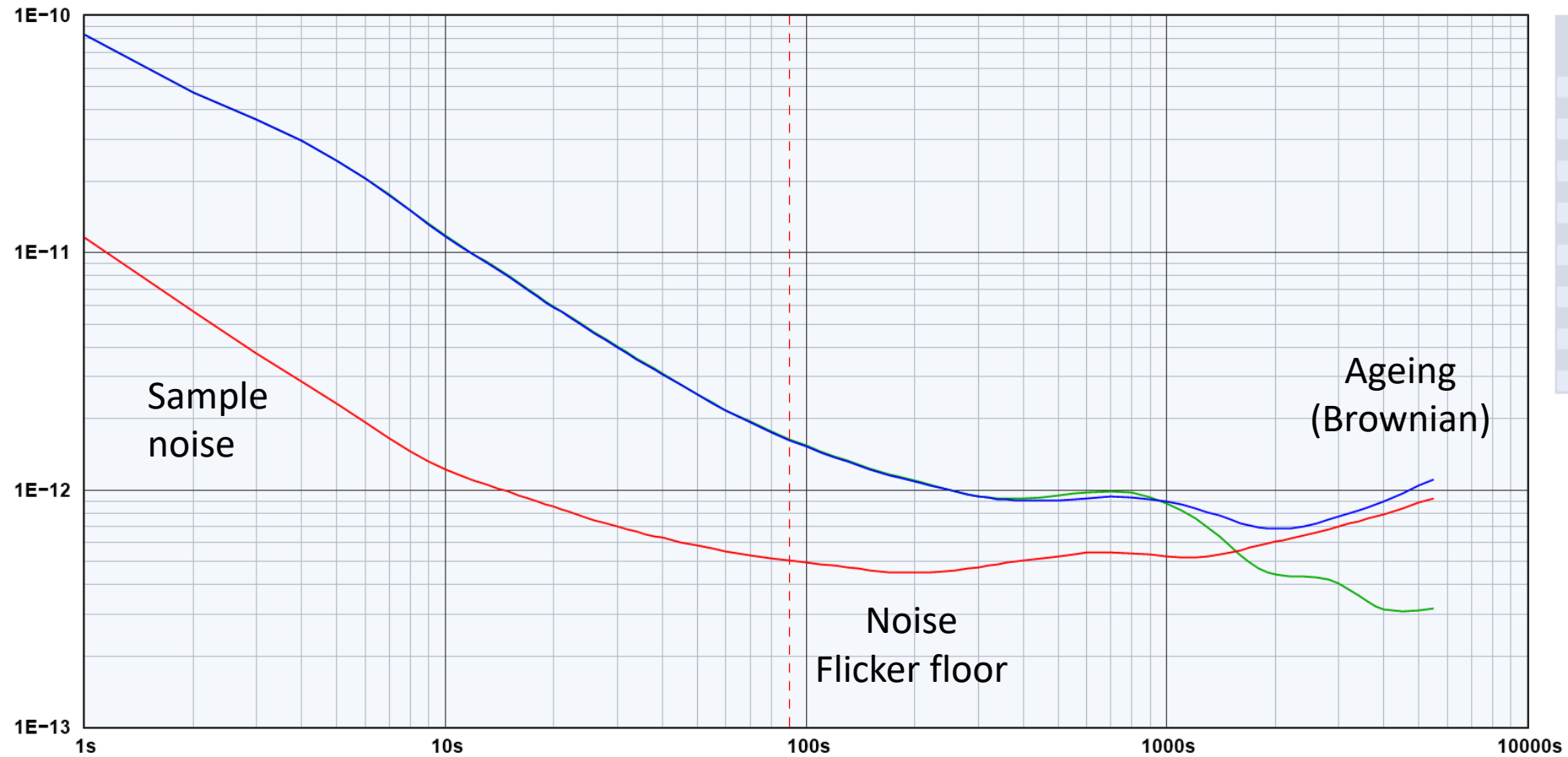
Oscillator model:

- 1: Sampling error
- 2: Noise flicker floor
- 3: Ageing effects

# Result of the lab experiment

TimeLab V1.77 (Beta) (x64) - C:\Users\eschr\Desktop\tinyGTC prototype\v01494\8h\P1 gps P2ocxo REF Rb ch 0.tim  
File Edit Trace Display Legend Measurement Masks Scripts Acquire Help

Modified Allan Deviation ( $\text{Mod } \sigma_y(\tau)$ )



Tau	Sigma(Tau)
1s	8.34E-11
2s	4.74E-11
4s	2.96E-11
8s	1.51E-11
10s	1.18E-11
20s	5.95E-12
40s	3.09E-12
80s	1.77E-12
100s	1.54E-12
200s	1.10E-12
400s	9.21E-13
800s	9.74E-13
1000s	8.81E-13
2000s	4.41E-13
4000s	3.14E-13

Trace	Notes	Input Freq	Sample Interval	MDEV at 90s	Duration	Elapsed	Acquired	Trace Window	Channel
P1 gps P2ocxo REF Rb	tinyGTC FW V0.1494	10 MHz	1 s	1.64E-12	8h	8h	28800 pts	28800 pts	Ch 0
P1 gps P2ocxo REF Rb	tinyGTC FW V0.1494	10 MHz	1 s	5.04E-13	8h	8h	28800 pts	28800 pts	Ch 1
P1 gps P2ocxo REF Rb	tinyGTC FW V0.1494	10 MHz	1 s	1.63E-12	8h	8h	28800 pts	28800 pts	Ch 2



# Field set-up



## Elements in this experiment

- OSC<sub>A</sub> is a transmitting beacon
- OSC<sub>B</sub> is a reference for a receiver
- Phase and frequency are transferred through a medium between A and B
- Phase definition is transferred by means of a modulation technique

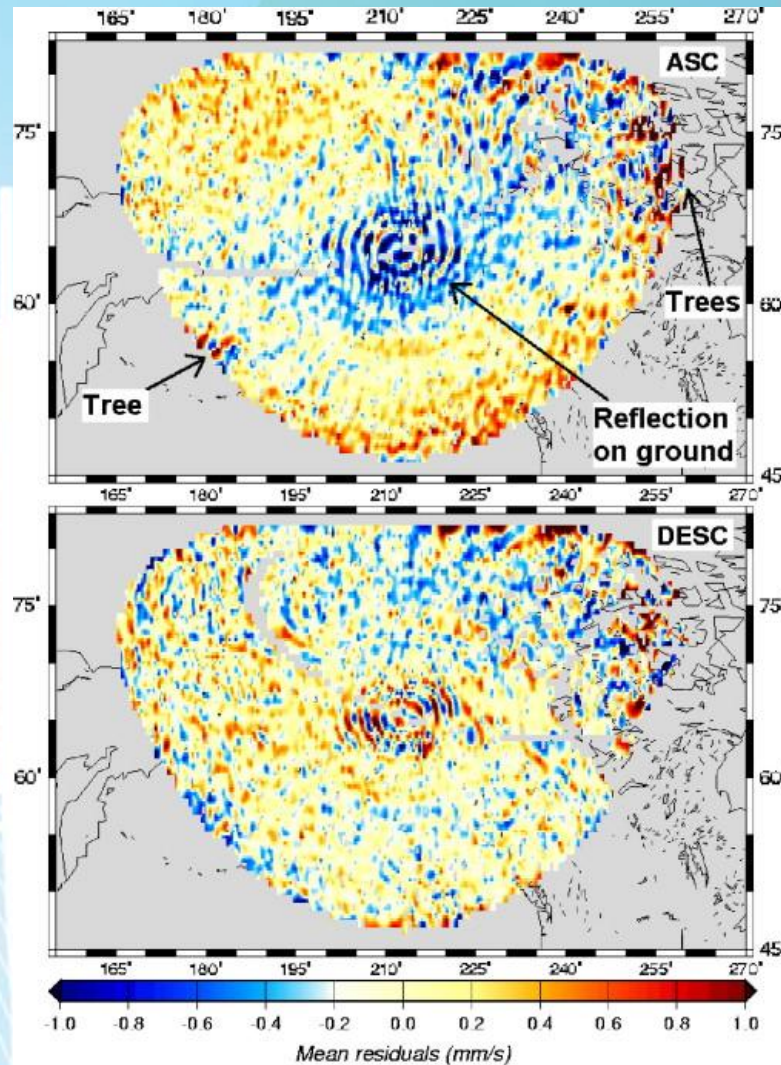
# Difference lab and field set-up

- Field set-up is affected by
  - Antenna's
  - Refraction
  - Geometry
- Antenna's:
  - Phase center definition depends on the design
  - Offset depending on topocentric azimuth and elevation
- Refraction
  - Wet and dry troposphere
  - Ionospheric delay
- Geometry
  - Relative velocity and distance
  - Light time effect
  - Time synchronization

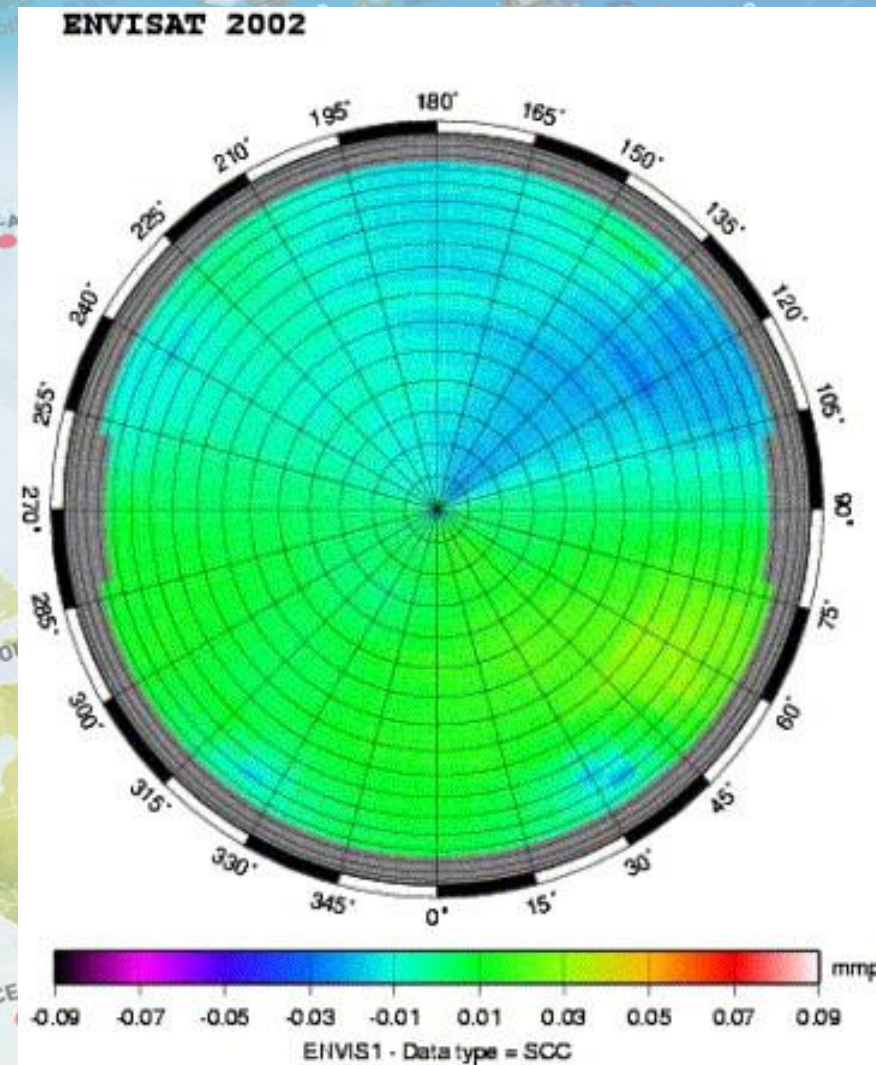




# Antenna phase center variations



POE orbit mean residuals of Fairbanks-SPOT-5 link from 2007/05/21 to 2007/08/28. P Yaya and C Tourain 2010 ASR



Willis, Desai, Bertiger, Haines and Auriol 2005 ASR



# Refraction

- In its most general form:
  - $\Delta s = \int_A^B (n - 1) ds$
  - Where the integral goes over the path A to B
  - Where  $ds$  is a small part along the path
  - Where  $n$  is an in-situ refractive index of the medium
- Troposphere : gasses between both antenna's
  - Non-dispersive effect so frequency independent
  - Wet troposphere : water vapor pressure (humidity)
  - Dry troposphere : dry gas content (air pressure)
  - With DORIS you estimate tropospheric parameters
- Ionosphere : free electrons cause the refraction



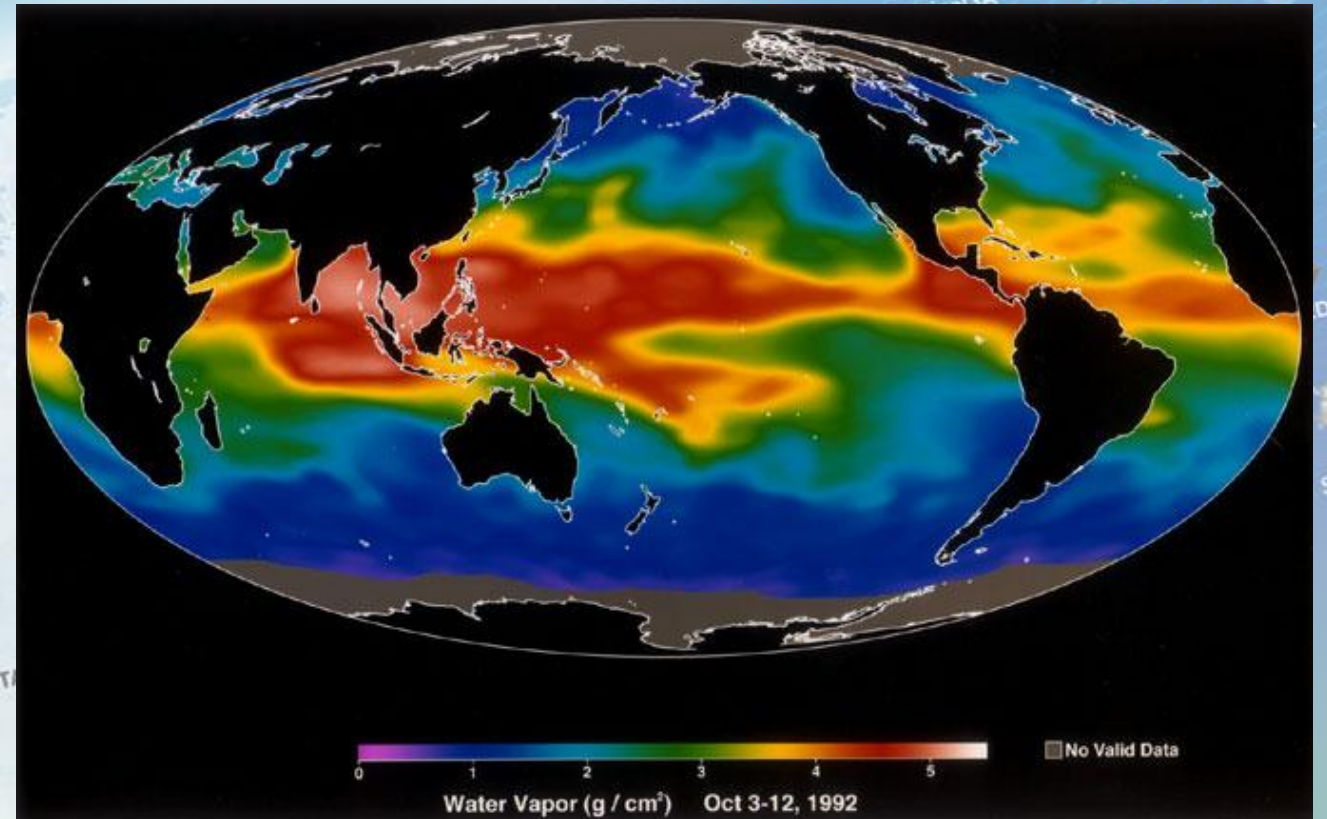
# Wet troposphere mapped with a microwave radiometer

Since the non-ionizing atmosphere is non-dispersive you need to resort to more difficult techniques to measure it. Most people don't do this, but for satellite altimeter systems we do.

The atmosphere is opaque on certain frequencies due to water vapor and this is what you can measure at the water vapor absorption line (22 GHz) relative to a few neighboring frequencies.

A radiometer is a noise receiver, it measures the thermal noise of the object, and hence its blackbody temperature.

(Radiometer: it is >10 million USD instrument)



# Ionospheric refraction

- Observe it with at least two frequencies (dispersive)

$$r(f_1) = r_o + \frac{\alpha}{f_1^2} \quad \text{and} \quad r(f_2) = r_o + \frac{\alpha}{f_2^2}$$

- Two equations with two unknowns, solve  $r_o$  and  $\alpha$
- The  $\alpha$  parameter results in the free electron content along the path between A and B, converts to TEC
- And  $r_o$  is ofcourse what we want in further processing
- One problem: the phase centers for  $f_1$  and  $f_2$  do not match for any antenna, there is a phase offset.

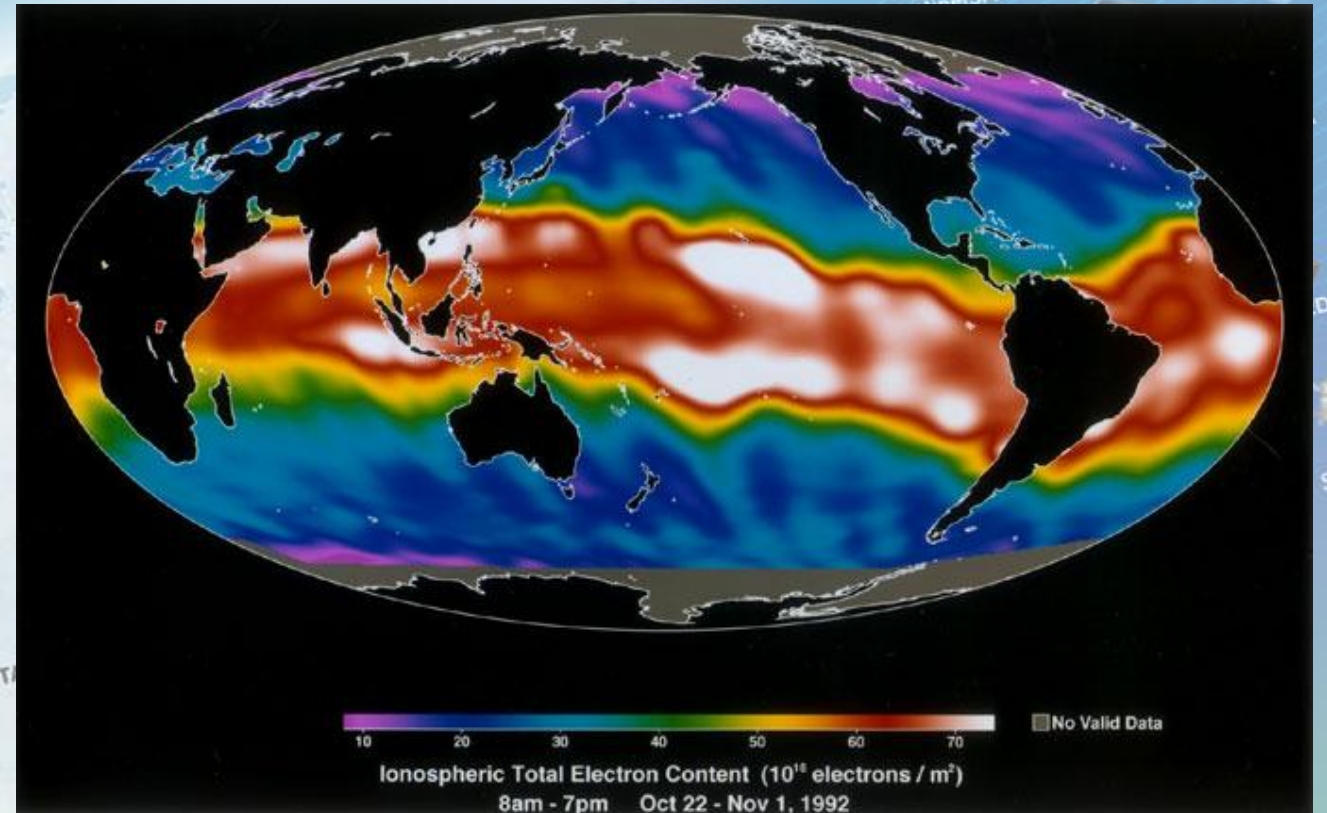


## Ionosphere mapped with dual frequency altimeter radar on C and Ku band

In this case you have a downward looking ocean sensing radar, it works on two microwave frequencies and you solve for the TEC which is downward to the sea surface.

The wavy red band coincides with the geomagnetic equator, this is where you can expect turbulence in the ionosphere.

Locally the ionosphere is a smooth function of time and place, we correct it with the DORIS data.



# Geometry: LOS velocity and distance

- Distance to phase relation  $\text{OSC}_A$  and  $\text{OSC}_B$

$$d = c(t_B - t_A) = \varphi_B - \varphi_A$$

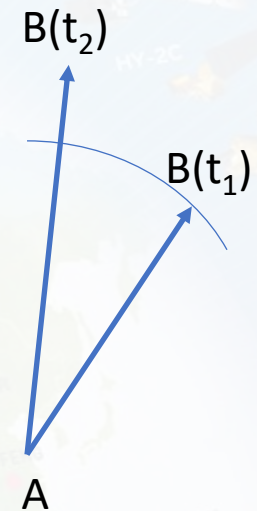
- Line of sight velocity  $\text{OSC}_A$  and  $\text{OSC}_B \Rightarrow$  Doppler signal

$$f_o + \Delta f = 1 + \left(\frac{\mathbf{v} \cdot \mathbf{n}}{c}\right) f_o \quad : \quad \mathbf{v} \text{ velocity and } \mathbf{n} \text{ direction}$$

- Integrate  $\Delta f$  over time and you get  $\varphi(t_1, t_2)$  along  $\mathbf{n}$

$$\varphi(t_1, t_2) = \int_{t_1}^{t_2} \lambda \Delta f dt$$

- This is the phase difference between  $t_1$  and  $t_2$  when  $c = \lambda f$  We also call it the observed carrier phase signal
- During POD  $\varphi(t_1, t_2)$  is used to estimate  $\varphi_B - \varphi_A$  for all defined epochs





# Light time correction

- The **time-of-flight** (or **light-time**) principle applies
  - To all measurements between the ground and the satellite
  - With **two-way** measurements we measure the round-trip time, thus a **pulse** leaves the laser, **reaches** the satellite where it **reflects**, and returns to a **detector** on the ground
  - With one-way measurements we measure the **one-way** trip from transmitter to receiver
- First and second order clock errors:
  - There is a **first-order** clock error  $c \cdot dt$
  - There is a **second-order** clock error  $v_{\text{sat}} \cdot dt$
- Both effects are significant
- Light time effect  $\neq$  Relativity (not in this lecture)

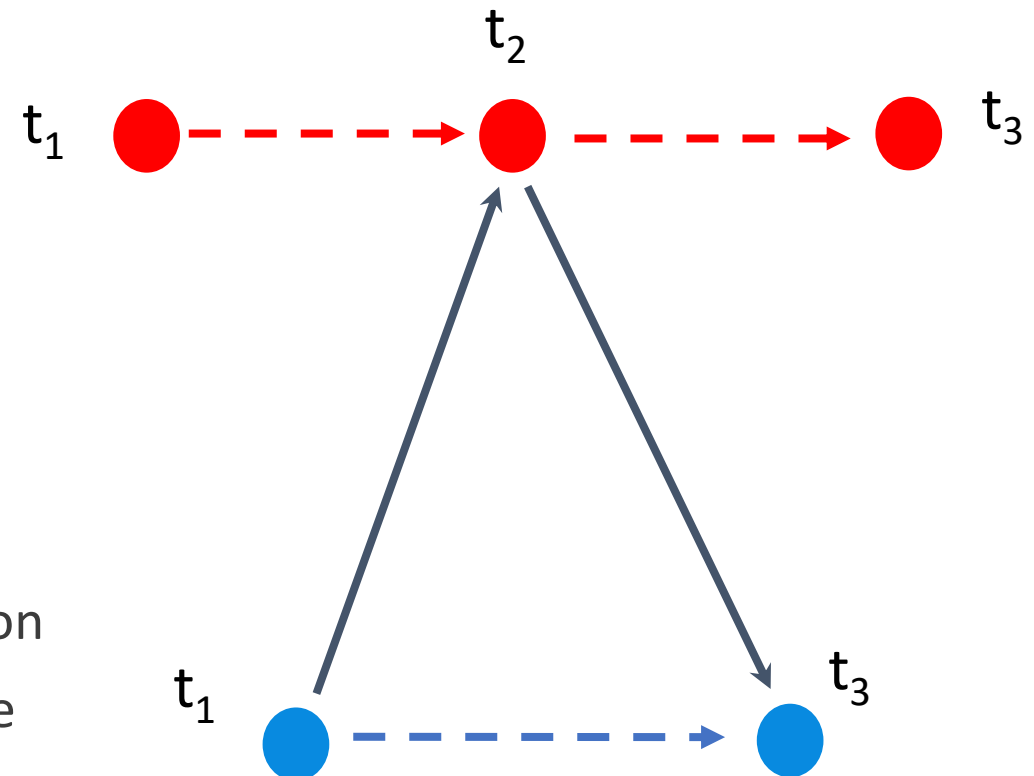
# Light-time effect illustrated with 2-way ranging

$t_1$  : laser sends the light pulse

$t_2$  : light bounces of satellite

$t_3$  : photons return in detector

- red is the satellite
- blue is the ground station
- black arrow is a light ray
- blue dashed arrow is motion ground station
- red dashed arrow is motion of the satellite





# Light time effect

Transmitter	Altitude (km)	Milliseconds
GRACE	450	1.50
CryoSat-2	700	2.33
Jason-3	1330	4.46
Lageos	5893	19.6
GPS	20000	86.73
Geostationary	36000	140.10
Moon	384400	1281.3

# Time synchronization

- On the satellite there is well behaved local oscillator (LO) with an ADEV like  $5 \times 10^{-13}$  at 1000s and  $10^{-11}$  at 1s. The LO needs to be modelled relative to a ground beacon in DORIS where the timing pulses are synchronized to TAI
- Master beacons that are synchronized to TAI, they transmit synchronized  $C_1$  and  $C_2$  codes. So we have two tasks:
  - Task 1: Model the LO on the satellite with  $C_1$  and  $C_2$  which can only be done during orbit determination (OD)
  - Task 2: On non-master beacons you have an independent LO that needs to be modelled as well, also during OD either via beacon biases or more sophisticated techniques
- A part of the work is already done for you in the RINEX files where the epochs are in principle non-synchronized LO derived, but there is a correction field.



# Differences to GNSS

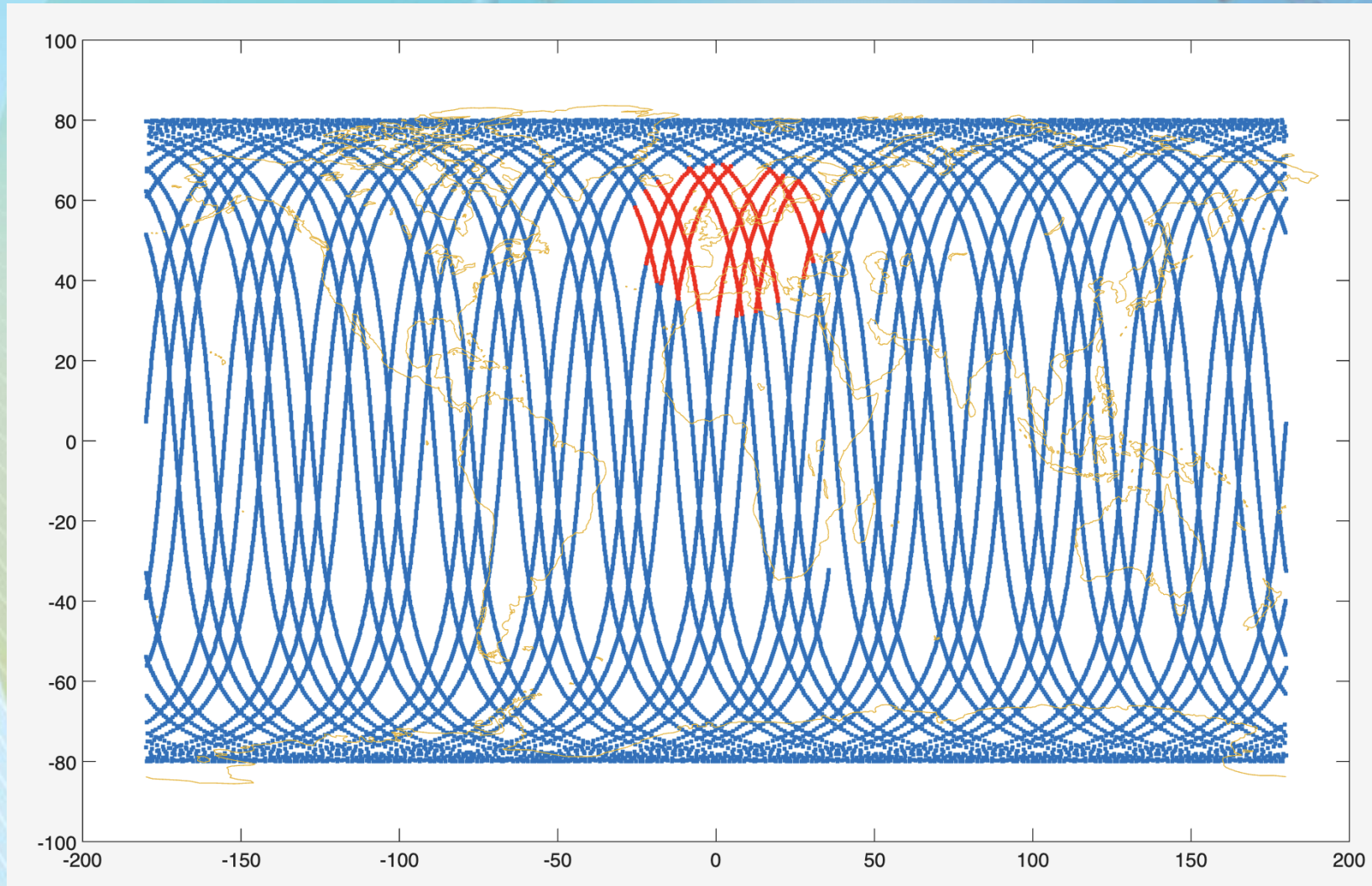
- Similar : Pseudoranges and carrier phase on 2 frequencies
- DORIS frequencies are further apart: 400 MHz and 2 GHz
- No redundancy with pseudoranges, no easy navigation solution
- There is a DORIS navigator on the satellite, it is called DIODE
- DORIS receiver collects carrier phase integrated over 10 seconds from the beacons it encounters
- The data is downloaded and disseminated to the users.
- Modern way of transfer is the RINEX format.

# Demonstration of navigation with DORIS

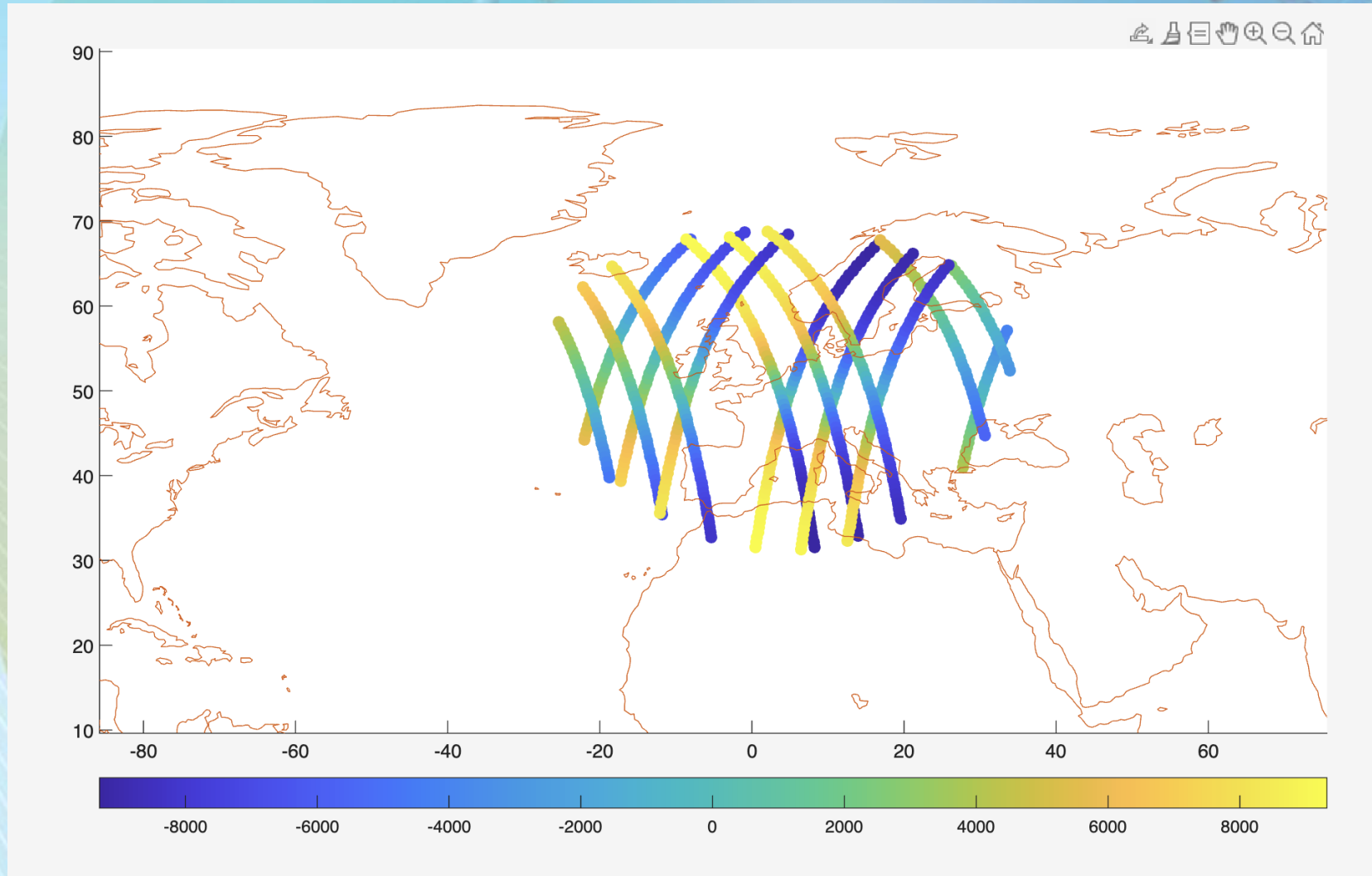
- Assume for simplicity that the satellite's orbit is known and that the satellite carries a receiver
- There is one transmitting beacon in the field,
- Beacon: unknown position, small frequency offset
- Observe only the Doppler effect in space
- Now estimate the beacon position and the beacon frequency offset from the satellite data from a known orbit.
- Go to the argos simulator and your Jupyter notebook and play around with the program.



# Beacon seen along tracks

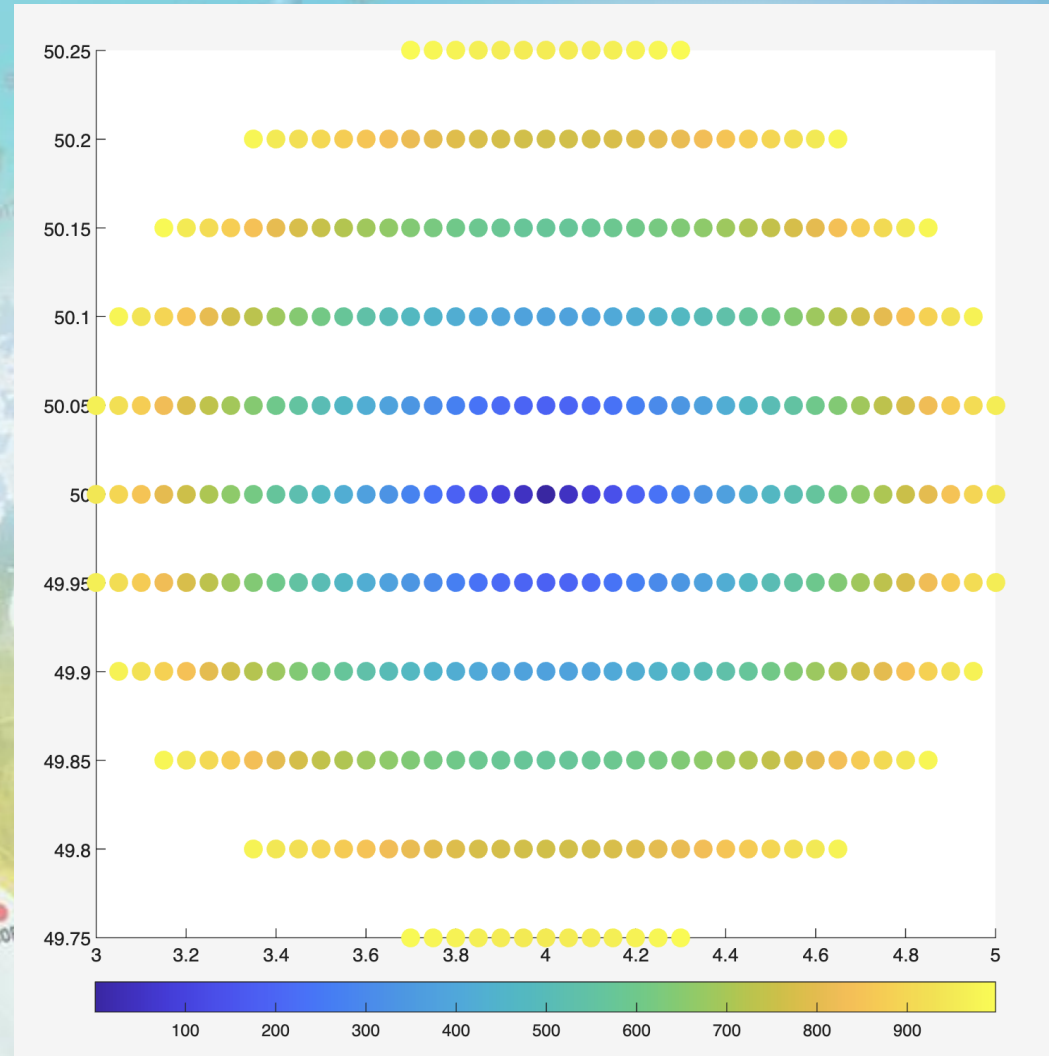


# Doppler seen by satellite





# Residuals grid search method



# Key points

- Point #1  
The twin oscillator experiment in the laboratory results at best in an Allan deviation analysis
- Point #2  
Field experiment differs from lab because of Antenna, Refraction and Geometry effects
- Point #3  
Antenna: phase center offsets  
Refraction: Wet/Dry troposphere and Ionosphere
- Point #4  
We need to deal with a light time correction
- Point #5  
We demonstrated the Doppler frequency to carrier phase relation
- Point #6  
We explained time synchronization in DORIS and discussed the differences between GNSS and DORIS
- Point #7  
Demonstration of a Doppler Navigation example by using a Jupyter notebook





Any questions so far?